# **PAPER • OPEN ACCESS**

# Deflection of Thin-Walled Panels Loaded in Shear with Different Types of End Stiffeners

To cite this article: E A Mousa et al 2020 IOP Conf. Ser.: Mater. Sci. Eng. 974 012012

View the article online for updates and enhancements.

# You may also like

- Numerical study on duplex stainless-steel plate girder under simultaneous bending and shear T C Sajith and M S Ajith
- Experimental Study on Shear Strength Behaviours of the Composite UHPC Steel Girders in Slender web Haitham H. Muteb and Dhafer M. Hasan
- Analysis of the Shear Angle in Corrugated Web Girders Witold Basinski



# 245th ECS Meeting San Francisco, CA

May 26-30, 2024

**PRIME 2024** Honolulu, Hawaii October 6-11, 2024

Bringing together industry, researchers, and government across 50 symposia in electrochemistry and solid state science and technology

Learn more about ECS Meetings at http://www.electrochem.org/upcoming-meetings



Save the Dates for future ECS Meetings!

13th International Conference on Civil and Architecture Engineering (ICCAE-13) **IOP** Publishing IOP Conf. Series: Materials Science and Engineering 974 (2020) 012012 doi:10.1088/1757-899X/974/1/012012

MilitaryTechnicalCollege Kobry El-Kobbah, Cairo, Egypt



13<sup>th</sup> International Conference on Civil and Architecture Engineering **ICCAE-13-2020** 

# **Deflection of Thin-Walled Panels Loaded in Shear with Different Types of End Stiffeners**

E A MOUSA<sup>1</sup>, M H EL-BOGHDADI<sup>2</sup>, N M YOSSEF<sup>2,3</sup>

<sup>1</sup>PhD student, department of Structural Engineering, Faculty of Engineering, Tanta University, Tanta,

<sup>2</sup>Associate Professor, Dept. of Structural Engineering, Faculty of Engineering, Tanta Univ., Tanta,

<sup>3</sup> Corresponding author (Mobile: +201225151616; Fax: +20403315860; Email: nashwa.yossef@gmail.com, nashwa\_abdeltawab@f-eng.tanta.edu.eg

Abstract, Numerical analysis for the load/deflection behavior of thin-web plate girders have been studied in this paper using a finite element program. An analytical approach presented in literature was discussed, and hence, it was used to calculate the load/deflection values in prebucking and post buckling stages. The validated finite element model is used to discuss the meaningful of the analytical approach. The effect of different parameters on load-deflection curves is proposed. Stiffener thickness  $(t_s)$ , end post type (no end post (NEP), rigid end post (REP), and non-rigid end post (NREP)) and end distance (e) are the main parameters considered in this paper. The analytical approach estimates well the load/deflection behaviors of REP and NREP plate girders with  $(a/h_w < 2.2)$ . On the other hand, the analytical approach can't estimate either for the shear strength or the deflection of NEP plate girders.

# **KEYWORDS**

Plate girders; Shear behaviour; In-plan deflection; End stiffeners; Finite element model.

# 1. INTRODUCTION

Plate girder is a structural element which contains steel plates assembled together by welded or bolted to one another. Rolled sections account for structures with relatively low loads and limited spans. However, when it comes to relatively heavy loads with large spans, as bridge girders with small to medium spans, designed plate girders are among the most widely used sections [1, 2].

Optimizing the plate girder dimensions presents deep thin web, that thin web tends to buckle. Hence transverse intermediate stiffeners are used to restrain the web and increase the buckling strength. At shear force, the thin web panels buckle. Therefore the shear behavior of thin web panel is one of the important design aspects of plate girder [3, 4].

A thin web panel subjected to shear was studied by several researchers [3-14]. The shear resistance of transversely stiffened plate girders was affected mainly by three factors [15]; 1) the buckling shear resistance, 2) the post-buckling shear resistance, and 3) flanges contribution to the shear resistance.

The buckling shear resistance was studied and formulated by Porter et al. [16], hence Lee and Yoo [4] propose design equations for the determination of ultimate shear strength of the web panels. In the meanwhile, Ajam and Marsh express the elastic shear buckling strength based on the Tresca yield

Content from this work may be used under the terms of the Creative Commons Attribution 3.0 licence. Any further distribution of this work must maintain attribution to the author(s) and the title of the work, journal citation and DOI. Published under licence by IOP Publishing Ltd 1

**IOP** Publishing

IOP Conf. Series: Materials Science and Engineering 974 (2020) 012012 doi:10.1088/1757-899X/974/1/012012

criterion [17]. For all studied models, the shear buckling coefficient is the key parameter for the shear strength. As a simplification, and as an accurate approximation, all panel edges are assumed to be simply supported, although Galambos has present the shear buckling coefficient for panel with fixed rotational restraints at the top and bottom [18].

Wilson in 1886 was one of the first to discuss the post-buckling behaviour [19]. In a major advance in 1950, Basler and Thurlimann [20] presented extensive studies on the post-buckling behaviour of plate girder web panels. They assumed that the transverse stiffeners doing as anchors and the yield was far from flanges. Then later Basler [21, 22] presented different approach on the post-buckling behaviour of thin web panel of a plate girder. In 1970s, Rockey and his co-workers predicted the ultimate strength of webs subjected to shear [16, 23].

As noted by many researchers, increasing flange dimensions develops larger shear strength, which means flanges contribute the shear strength by the frame action. However, Höglund [24] calculated the shear contribution from the frame action.

A numerical study of nonlinear large deflection behaviour of plate girder under shear was presented by Alinia et al. [7]. It was observed that plastic hinges induced by shear in the end panel only. These hinged are induced by the shear deformation near the supports. The researchers state that the end-posts are important elements in the behaviour and design of plate girder under shear.

A number of experimental and analytical studies have been carried out to establish a mechanical model to represent the out-of-plane and the in-plane deformation characteristics of web panel under shear [25, 26]. However the proposed model by Qian and Tan [25] does not take the effect of end-post into account, there is still a need for general analysis and design methodology to combine the effect of endpost on the out-of-plane and the in-plane deformation characteristics of web panels under shear.

Hence, this paper focuses on the shear behaviour of the thin web panel of plate girders. The principal aim is to improve the current presented model of the structural behaviour for the thin panel under shear formed by Qian and Tan [25]. The key parameters that affect this panel in the current investigation are the stiffeners thickness  $(t_s)$ , the type of end-post, the web aspect ratio  $(a/h_w)$  and the end-post distance (e). The scope of this study can be summarized as:

- 1. Discuss the current literature model for calculating the in-plane deflection of web panels loaded in shear.
- 2. Propose and validate a detailed finite element 3D model used to address the behaviour of fullscale plate girders.
- 3. Investigate factors affecting the in-plane deflection, and compare between the finite element results and current literature model.

#### 2. LITERATURE APPROACH FOR IN-PLANE DEFLECTION

Qian and Tan [25] presented a mechanical approach to calculate the in-plane deflection of web panel loaded in shear. Qian had used Cardiff theory to calculate the shear strength in his analytical approach. In Qian approach the overall deflection is divided into two stages, which are deflection of pre-buckling stage ( $\delta_{cr}$ ) and deflection of post-buckling stage ( $\delta_b$ ). Such that the total deflection for the panel loaded in shear  $(\delta_{pb})$  could be calculated as follow;

$$\delta_{pb} = \delta_{cr} + \delta_b$$

Thus, the pre-buckling deflection can be calculated as:

$$\delta_{cr} = 2 \tau_{cr} a (1+v)/E$$

in which: *a* is the panel length; v is the Poisson's ratio; *E* is the elastic modulus;

 $\tau_{cr}$  is the shear buckling stress of the web panel  $\left( = K \frac{\pi^2 E}{12 (1-\nu^2)} \left( \frac{t_w}{h_w} \right)^2 \right);$ 

2

1

13th International Conference on Civil and Architecture Engineering (ICCAE-13) **IOP** Publishing IOP Conf. Series: Materials Science and Engineering 974 (2020) 012012 doi:10.1088/1757-899X/974/1/012012

K is the shear buckling coefficient in case of web panel with transversal stiffeners only and equals to  $(5.34 + 4/(a/h_w)^2)$  $a/h_w \ge 1.0$ 

$$\begin{cases} 4 + 5.34/(a/h_w)^2 & a/h_w < 1.0 \\ \text{In the meanwhile, the post-buckling deflection was derived for the shear case as follows:} \\ \delta_b &= \frac{(\sigma_t^y)^2 c^2 h_w t_w}{E(V_{ult}c^{-4} M_{pf})} \end{cases}$$

$$3$$

where:

c is the distance assumed between the plastic hinge and the end bearing stiffener

$$\left(= a \left(0.25 + \frac{1.6 b_f t_f^2 f_{yf}}{t_w h_w^2 f_{yw}}\right)\right)$$

 $M_{pf}$  is the plastic moment capacity of flanges (=  $0.25 f_{yf} A_f t_f$ )  $\sigma_t^y$  and  $V_{ult}$  are the tensile yield stress of web panel under shear and the post-buckling shear force component, respectively, which can be calculated as:

$$\sigma_t^{\mathcal{Y}} = -\frac{3}{2}\tau_{cr}\sin 2\theta + \sqrt{f_{\mathcal{Y}W}^2 + \tau_{cr}^2[(1.5\sin 2\theta)^2 - 3]}$$
<sup>4</sup>

$$V_{ult} = \sigma_t^{y} h_w t_w \sin^2\theta \left(\cot\theta - \frac{a}{h_w}\right) + 4 \sigma_t^{y} t_w \sin\theta \sqrt{\frac{M_{pf}}{\sigma_t^{y} t_w}}$$
 5

Thus,  $\theta$  is the inclination of the tensile field, the angle  $\theta$  may be calculated by iteration to give the maximum value of  $V_{ult}$  or can be approximated as  $\left(\theta = 2/3 \tan^{-1}(\frac{h_w}{a})\right)$  as given by Rockey et al. [24].

However, equation 5 is subject to the following control [27]:

$$\frac{M_{pf}}{h_w^2 t_w f_{yw}} < 0.125 \left(\frac{h_w}{a}\right)^2 \left\{ -\frac{3}{2} \frac{\tau_{cr}}{f_{yw}} + \sqrt{1 - 0.25 \left(\frac{\sqrt{3}\tau_{cr}}{f_{yw}}\right)^2} \right\}$$

If the left-hand side is greater than the right-hand side stated in equation 6, the value of  $V_{ult}$  is given by:

$$\frac{V_{ult}}{\tau_{cr}t_{w}h_{w}} = 4\sqrt{3}\left(\frac{h_{w}}{a}\right)\frac{M_{pf}}{h_{w}^{2}t_{w}f_{yw}} + \left\{-\frac{3\sqrt{3}}{4}\frac{\tau_{cr}}{f_{yw}} + \sqrt{0.75 - \frac{3}{16}\left(\frac{\sqrt{3}\tau_{cr}}{f_{yw}}\right)^{2}}\right\}$$

#### 3. FINITE ELEMENT MODEL

A detailed finite element (FE) 3D model used to address the behaviour of transversely stiffened, fullscale plate girders was developed and verified. Nonlinear behaviour of the thin web panel was presented. The verified FE model is used to promote the current literature equations for in-plane deflection and study additional factors affecting the behaviour of thin web panels under shear force.

#### 3.1 Model description

Transversely stiffened plate girders were modelled using reduced-integrated shell element S4R. The element S4R, from non-linear finite element analysis of the ABAQUS platform element library [28], has four nodes each with six degrees of freedom. The full-scale 3D model includes both geometric and material nonlinearity. The mesh convergence study verified that the element with size 30 mm x 30 mm gives reasonable accuracy.

Girders with no end post (NEP), with rigid end post (REP), and with non-rigid end post (NREP) were included. Plated girder dimensions notations and various end supports were introduced, according to EN 1993-1-5 [29], as shown in figure 1.

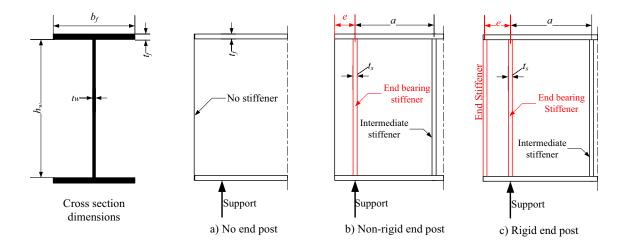


Figure 1. Cross section dimensions and types of end supports.

# 3.2 Boundary conditions and load procedures

The boundary conditions and load procedures follow what is indicated in [7], to assure quite small bending moment and constant shear force for the end panel. However, two-point loads were applied at the third points, and simply supported boundary conditions were applied to the end section. Both load and boundary conditions are presented in figure 2.

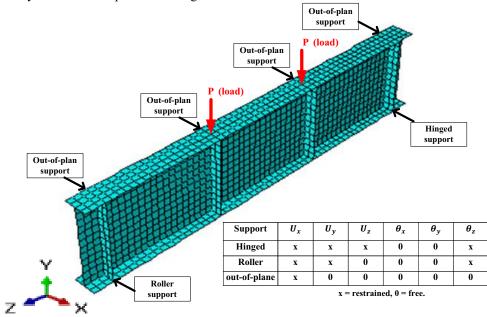


Figure 2. FE mesh, load and boundary conditions used.

# 3.3 Material properties

Bilinear elastic-plastic curve with linear strain hardening was used to simulate the behavior of the steel material for web, flanges and stiffeners. The slope of the linear elastic part of the curve, which mean

Young's modulus, was used as E = 200 GPa, and Poisson's ratio was taken v = 0.3. The slope of the hardening part was taken as 2 GPa. The used steel material is S355, which has ultimate stress 510 MPa and yield stress 355 MPa.

#### 3.4 Model validation

To determine the accuracy of the FE model, the model was used to simulate the behaviour of specimens exhibited in the experimental tests of Zhu and Zhao [26]. As well as, the numerical model studied by Alinia et al. [7]. Table 1 presents the comparison between the available results in the literature and the FE model results. The mean value for the ratio of the available ultimate load in literature ( $P_u$ ) to the FE ultimate load ( $P_{u,FE}$ ) is 0.97 with standard deviation of 0.05.

The failure modes of plate girders in [26] were compared with the failure of FE model, as shown in figure 3, which indicates a good agreement. In the meanwhile, the failure modes stated by [7] for studied plate girders were also confirmed by the FE study, as shown in figure 4. Finally, the comparisons between load-deflection curves for the literature plate girders and the FE load deflection curves are presented. Where the FE model accurately predicts the load-deflection curves, as can be seen in figure 5.

Table 1. Comparison between available ultimate load in the literature $(P_{u,L})$ and the	the FE one $(P_{u,FE})$ .
---	---------------------------

			Deta	ils of exp	periment	al tests c	of Zhu and	Zhao [26]			
Plate	$h_w$	tw	$b_{\mathrm{f}}$	$t_{\mathrm{f}}$	L	а	f <sub>vw</sub>	Е	$P_{u,L}$	$P_{u,FE}$	P <sub>u,L</sub> /
girder	[mm]	[mm]	[mm]	[mm]	[mm]	[mm]	[MPa]	[GPa]	[kN]	[kN]	$P_{u,\mathrm{FE}}$
G1	600	8	180	16	2700	900	280	208	1200	1291	0.93
G2	600	8	180	16	3600	1200	280	208	985	1037	0.95
		De	tails of t	he simu	lated pla	te girder	studied b	y Alinia et	al. [7]		
G(A)	1000	4	300	9	4000	1000	345	210	800	773	1.03
G(B)	1000	3.33	300	7.5	4000	1000	345	210	N/A	551	N/A
G(C)	1000	3.33	300	3.33	4000	1000	345	210	N/A	312	N/A
						Mean*					0.97
Standard deviation 0									0.053		

\* The ultimate load  $P_{u,L}$  for the plate girder G(B) and G(C) are unavailable, hence they were excluded from the mean and the standard deviation

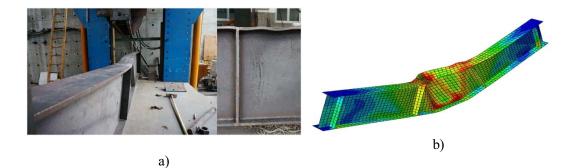


Figure 3. Lateral torsional buckling of tested specimen G2: a) Experimental results [26], b) F.E. results

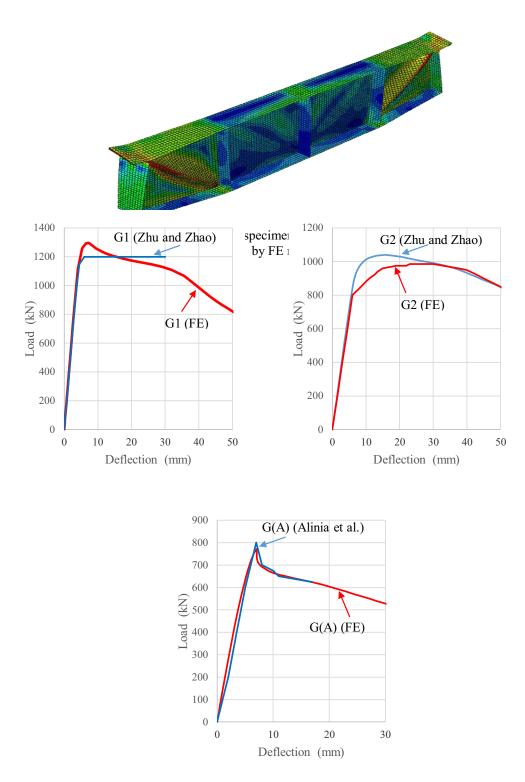


Figure 5. Comparison between literature girders and FE load-deflection curves.

# 4. NUMERICAL STUDY

# 4.1 Assumptions and parameters

In this section, the analytical and numerical load/deflection behaviour of thin-walled panel loaded in shear is investigated. Based on the main target of that section for studding the shear failure, the span length of the modelled plate girders was taken as 3 times the panel width (3*a*), so the bending effect on the studied panel was minimized. Regarding the geometrical limitation of thin-walled panel,  $h_w/t_w$  ratio was taken as 200 for all studied panels. Twenty ( $a/h_w$ ) ratios were investigated, such that ( $a/h_w$ ) ratio was varied from 3.5 to 0.8. The initial imperfections values ( $w_o$ ) were introduced for the model with scale factors ( $h_w/200$ ), which considers a greater value for the bow of the web.

The current study contains four hundred full scale thin-web plate girders, as shown in table 2, covering three main parameters; stiffener thickness ( $t_s$ ), end-post type (REP, NREP and NEP) and end distance (e).

Parameter	No. of	$h_w$	$\mathbf{a}/h_{w}$	$t_w$	$h_{w'}/t_{w}$	span	$p_f$	tf	End stiffeners type	Initial	$t_s$
	specimens	(mm)		(mm)		(T)	(mm)	(mm)		imperfection	( <i>mm</i> )
						(mm)				Wo	
Effect of	160	1000	3.5, 3.4, 3.3, 3.2,	5	200	3a	300	25	Rigid end post (REP),	$h_w/200$	5
stiffener			3.1, 3.0, 2.8, 2.6,						Non-rigid end post		10
thickness (t <sub>s</sub> )			2.4, 2.2, 2.0, 1.8						(NREP)		15
			1.7, 1.6, 1.5, 1.4,								20
			1.3, 1.2, 1.0, 0.8								, 
Effect of end-	60	1000	3.5, 3.4, 3.3, 3.2,	5	200	3a	300	25	No-end post (NEP),	$h_w/200$	10
post			3.1, 3.0, 2.8, 2.6,						Rigid end post (REP),		
			2.4, 2.2, 2.0, 1.8						Non-rigid end post		
			1.7, 1.6, 1.5, 1.4,						(NREP)		
			1.3, 1.2, 1.0, 0.8								
Effect of end	180	1000	3.5, 3.4, 3.3, 3.2,	5	200	3а	300	25	e = 100  mm for NEP,	$h_w/200$	10
distance ( <i>e</i> )			3.1, 3.0, 2.8, 2.6,						ALLE, INALE		
			2.4, 2.2, 2.0, 1.8						e = 150 mm tor NEP, REP, NREP		
			1.7, 1.6, 1.5, 1.4,						e = 300  mm for NEP.		
			1.3, 1.2, 1.0, 0.8						REP, NREP		

IOP Conf. Series: Materials Science and Engineering **974** (2020) 012012 doi:10.1088/1757-899X/974/1/012012

8

#### 4.2 Results discussion

The specimens listed in table 2 were solved numerically using (ABAQUS), hence load-deflection curves for these specimens were determined numerically and analytically. Investigating the effect of different parameters on the finite element load-deflection curves, then conforming these results to the theoretical approach is the main goal of this study.

The results and the primary parameters are presented and discussed herein, where  $P_{u,FE}$  is the ultimate load carrying capacity calculated numerically using the FE model, which is the maximum load stated, and was typically followed by a drop in the load value.  $\delta_{Pb,FE}$  is the maximum FE vertical deflection under loading points for the post buckling stage, which is followed by failure stage (an increase in the deflection with nearly constant load).

In the meantime, the shear deflection  $\delta_{pb,a}$  calculated by the analytical approach were expressed in equation 1, and the ultimate shear force  $P_{u,a}$  is calculated such that;

 $P_{u,a} = \tau_{cr} \left( A_w \right) + V_{ult}$ 

8

### 4.3 Effect of stiffeners thickness

To investigate the effect of stiffeners thickness, 160 specimens were studied as listed in table 2. Specimens with REP and NREP were examined. The specimens had the same cross section dimensions, but different stiffeners thickness (5 mm, 10 mm, 15 mm, and 20 mm). Twenty ratios of  $(a/h_w)$  were studied separately and combined, based on the aspect ratio of the web panels to evaluate the elastic shear buckling strength and post buckling strength, [4].

Table 3 & 4 present the comparison between the analytical results of the presented approach and the FE results for girders with stiffeners thickness 5 and 20, respectively. When the thickness of stiffeners were 5 mm, the mean values of  $P_{u,a'}P_{u,FE}$  for REP and NREP are 0.96 and 1.00, with standard deviations of 0.17 and 0.19, respectively. Then the mean values  $P_{u,a'}P_{u,FE}$  are 0.9 and 0.95 with standard deviations 0.11 and 0.15, for REP and NREP girders respectively, with stiffeners thickness 20 mm. Which indicates a good agreement with the analytical results.

For girders with thin stiffener ( $t_s = 5$  mm), the mean values of  $\delta_{pb,a}/\delta_{pb,FE}$  for REP and NREP are 0.80 and 0.82, with standard deviations of 0.11 and 0.12, respectively. Hence for girders with thick stiffeners ( $t_s = 20$  mm), the mean values of  $\delta_{pb,a}/\delta_{pb,FE}$  for REP and NREP are 0.82 and 0.87 respectively, with standard deviations of 0.10, for both. Deflection values indicate slight discrepancies between the FE and analytical approach. The authors believe that the presented approach underestimate the deflection values, since it only considers the shear deflection. But the FE model were subjected to shear and bending effects. Furthermore, results confirmed the small impact of the stiffener thickness on both shear strength and deflection of thin girders.

Figures 6, 7 and 8 present comparisons between the analytical and the FE load-deflection curves for different specimens, where it can be seen that the analytical approach accurately predicts the load-deflection curves for the studied specimens, especially for girders with panel aspect ratio ranged from  $a/h_w = 1.7$  to  $a/h_w = 2.2$ , hence the performance has some discrepancy for  $a/h_w > 2.2$ . That was probably as discussed, such that for  $a/h_w > 2.2$ , the effect of bending was significant as the panel and the girder length increased. Although these slight deviations between analytical predictions and numerical results, the given approach still exhibits acceptable accuracy compared with these FE especially for  $a/h_w < 2.2$ . In the meantime, the stiffeners thickness has negligible effect on the load-deflection behavior.

		a		ts	Anal	ytical	FE (	REP)	FE (ľ	NREP)	Ra	tio	Ra	tio
(m			(m		E	q.					(RI	E <b>P</b> )	(NF	REP)
$\mathbf{h}_{\mathbf{w}}$ . $\mathbf{t}_{\mathbf{w}}$ ( $\mathbf{m}_{\mathbf{m}}$ )	a/ h <sub>w</sub>	mm	$\mathbf{b}_{\mathbf{f}} \mathbf{x} \mathbf{t}_{\mathbf{f}} (\mathbf{m}_{\mathbf{f}})$	mm	P <sub>u,a</sub> kN	$\delta_{pb,a}$ mm	P <sub>u,FE</sub> kN	δ <sub>Pb,FE</sub> mm	P <sub>u,FE</sub> kN	δ <sub>Pb,FE</sub> mm	$P_{u,a}\!/P_{u,FE}$	$\delta_{pb,a'}\delta_{Pb,FE}$	$P_{u,a}\!/P_{u,FE}$	$\delta_{pb,a'}\delta_{Pb,FE}$
	3.5	3500			701	19.14	954	23.20	939	23.00	0.73	0.83	0.75	0.83
	3.4	3400			714	18.12	950	21.41	843	21.16	0.75	0.85	0.85	0.86
	3.3	3300			727	17.12	952	19.84	942	19.74	0.76	0.86	0.77	0.87
	3.2	3200			742	16.15	961	24.28	946	23.90	0.77	0.67	0.78	0.68
	3.1	3100			757	15.21	969	22.40	951	22.26	0.78	0.68	0.80	0.68
	3.0	3000			773	14.30	976	20.64	952	18.36	0.79	0.69	0.81	0.78
	2.8	2800			808	12.57	987	15.37	962	13.04	0.82	0.82	0.84	0.96
1000 x 5	2.6	2600 2400	00 x 25	5	848 894	10.95 9.46	994 1005	12.53 12.73	973 971	12.62 11.40	0.85	0.87	0.87	0.87
100	2.2	2200	300		947	8.08	1005	10.91	989	10.52	0.94	0.74	0.96	0.77
	2.0	2000			1009	6.83	1027	8.29	992	10.01	0.98	0.82	1.02	0.68
	1.8	1800			1081	5.69	1083	6.73	1033	7.07	1.00	0.85	1.05	0.80
	1.7	1700			1123	5.16	1096	6.02	1032	5.71	1.02	0.86	1.09	0.90
	1.6	1600			1168	4.67	1095	5.07	1044	5.03	1.07	0.92	1.12	0.93
	1.5	1500			1217	4.20	1092	4.45	1063	4.88	1.11	0.94	1.14	0.86
	1.4	1400			1272	3.77	1135	3.98	1083	4.02	1.12	0.95	1.17	0.94
	1.3	1300			1332	3.36	1164	3.46	1106	3.50	1.14	0.97	1.20	0.96
	1.2	1200			1399	2.99	1193	4.42	1132	3.02	1.17	0.68	1.24	0.99
	1.0	1000			1556	2.33	1282	3.33	1203	3.36	1.21	0.70	1.29	0.69
	0.8	800			1769	1.82	1408	3.26	1314	3.53	1.26	0.56	1.35	0.52
							Mear	1			0.96	0.80	1.00	0.82
						Stan	dard de	eviation			0.17	0.11	0.19	0.12

Table 2 Analytical and EE loads/deflections of the studied on	$a_{a}$ and $a_{a}$ with $(t - 5 a_{a})$ we have
Table 3. Analytical and FE loads/deflections of the studied sp	becimens with $(I_{\circ} - J mm)$ values.
	(13)

		a		ts	Anal	ytical	FE (	REP)	FE (N	NREP)	Ra	tio	Ra	tio
(u			(u		E	lq.					(RI	E <b>P</b> )	(NR	REP)
h <sub>w</sub> . t <sub>w</sub> (mm)	a/ h <sub>w</sub>	mm	b <sub>r</sub> x t <sub>r</sub> (mm)	mm	$P_{u,a}$	$\delta_{pb,a}$	$P_{u,\text{FE}}$	$\delta_{Pb,F}$	$P_{u,\text{FE}}$	$\delta_{Pb,FE}$	m	н	m	Щ
w• tw	a		o <sub>f</sub> x t		kN	mm	kN	mm	kN	mm	$P_{u,a}\!/P_{u,FE}$	$\delta_{Pb, H}$	$P_{u,a}/P_{u,FE}$	$\langle \delta_{Pb,j}$
ĥ			_								$\mathbf{P}_{\mathrm{u,a}}$	$\delta_{pb,a'}\delta_{Pb,FE}$	$\mathrm{P}_{\mathrm{u,a}}$	$\delta_{pb,a'} ~ \delta_{Pb,FE}$
	3.5	3500			701	19.14	946	23.00	946	22.86	0.74	0.83	0.74	0.84
	3.4	3400			714	18.12	945	21.22	951	21.05	0.76	0.85	0.75	0.86
	3.3	3300			727	17.12	966	19.66	947	19.66	0.75	0.87	0.77	0.87
	3.2	3200			742	16.15	970	24.20	958	18.19	0.76	0.67	0.77	0.89
	3.1	3100			757	15.21	978	22.36	962	22.42	0.77	0.68	0.79	0.68
	3.0	3000			773	14.30	983	20.46	966	20.46	0.79	0.70	0.80	0.70
	2.8	2800			808	12.57	994	15.10	975	12.95	0.81	0.83	0.83	0.97
	2.6	2600			848	10.95	1002	12.91	985	11.52	0.85	0.85	0.86	0.95
	2.4	2400			894	9.46	1016	11.70	995	12.64	0.88	0.81	0.90	0.75
x 5	2.2	2200	: 25	20	947	8.08	1023	10.45	1018	10.25	0.93	0.77	0.93	0.79
1000 x 5	2.0	2000	300 x		1009	6.83	1054	8.26	1020	7.62	0.96	0.83	0.99	0.90
-	1.8	1800	63		1081	5.69	1119	6.65	1074	6.73	0.97	0.86	1.01	0.85
	1.7	1700			1123	5.16	1115	5.66	1077	5.70	1.01	0.91	1.04	0.91
	1.6	1600			1168	4.67	1151	5.00	1097	5.02	1.01	0.93	1.06	0.93
	1.5	1500			1217	4.20	1200	4.45	1124	4.47	1.01	0.94	1.08	0.94
	1.4	1400			1272	3.77	1251	3.92	1153	3.92	1.02	0.96	1.10	0.96
	1.3	1300			1332	3.36	1311	3.40	1185	3.41	1.02	0.99	1.12	0.99
	1.2	1200			1399	2.99	1387	4.32	1229	2.94	1.01	0.69	1.14	1.02
	1.0	1000			1556	2.33	1577	3.12	1351	3.13	0.99	0.75	1.15	0.74
	0.8	800			1769	1.82	1833	2.81	1497	2.17	0.97	0.65	1.18	0.84
							Mean	I			0.90	0.82	0.95	0.87
						Stan	dard de	viation			0.11	0.10	0.15	0.10

Table 4. Analytical and FE loads/deflections of the studied specimens with ( $t_s = 20 \text{ mm}$ ) values.

1800 1600 1400 1200 Load (2P) (KN) 1000 REP, ts=10 mm 800 REP, ts=20 mm 600 NREP, ts=10 mm 400 - NREP, ts=20 mm 200 --- Analytical eq. 0 0 5 10 20 25 15 Deflection under load (mm)

Figure 6. Comparison between analytical equation and FE load-deflection curves for REP and NREP plate girders with  $(a/h_w = 1.0)$ .

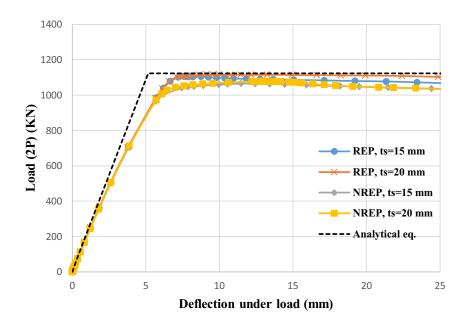


Figure 7. Comparison between analytical equation and FE load-deflection curves for REP and NREP plate girders with  $(a/h_w = 1.7)$ .

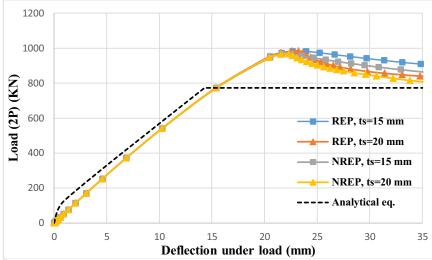


Figure 8. Comparison between analytical equation and FE load-deflection curves for REP and NREP plate girders with  $(a/h_w = 3.0)$ .

#### 4.4 Effect of end-post

Sixty plate girders having the same cross-sections and the same spans but with different end-post profiles were studied. Table 5 presents a comparison between analytical and FE load/deflection results for the studied plate girders. Since the analytical approach of Qian and Tan [25] neglected the effect of end-post, the analytical load/deflection results did not confirm any difference with different end-posts. Unlike the analytical results, the FE results highlight the effect of end-post rigidity on the load-deflection behaviour. The results show that providing a REP increases the ultimate shear strength ( $P_{u,FE}$ ) of the girder, if compared with NREP. In the meantime, changing end-post from REP to NREP has insignificant effect on maximum vertical deflection ( $\delta_{Pb,FE}$ ), in the most studied cases. It also shows that, ( $P_{u,FE}$ ) of the same plate girder considerably decrease by providing NEP. However NPE profile has irregular effect on ( $\delta_{Pb,FE}$ ).

The column charts presented in figure 9 shows the ratio between the analytical approach and finite element results for REP, NREP and NEP plate girders with  $a/h_w = 0.8$ , 1, 1.4, 1.7, 2, 2.2 and 3. The analytical approach estimates well the shear strength of REP and NREP plate girders, hence it can't estimate the shear strength or deflection for NEP plate girders.

In the meantime, the presented analytical approach gives underestimates deflection results for both REP and NREP plate girders.

#### 4.5 Effect of end-distance (e)

The current sub-section studies the effect of end-distance (e) in the load-deflection curves, note that the cross-section dimensions of the studied plate girders remained constant. However,  $a/h_w$  ratio was varied from 0.8 to 3.5, as listed in tables 5, 6 and 7 for (e) equals 100, 200 and 300 mm, respectively.

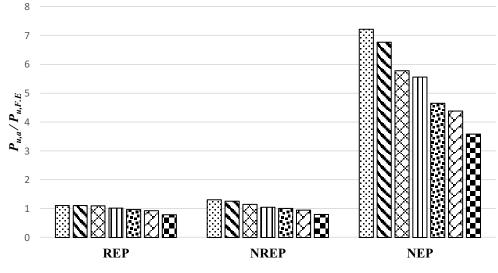
Figure 10 presents the load-deflection curves of plate girders  $(a/h_w = 2)$  with different end-post profiles. Accordingly, the end distance (*e*) are 100 mm, 150 mm and 300 mm. the FE results show that increasing the end-distance (*e*) has no effect on the initial stiffness of the plate girders.

However, it slightly increases the ultimate shear strength of the girders. That was expected, since increasing the end-distance influence the rigidity of end-post and hence the position of plastic hinges on flanges at failure stage. Hence, that explains the insignificant change of the maximum deflection at the post buckling stage for the studied girder. In other word, the end-distance becomes effective only in the failure stage. Figure 10 also states the misestimating of the analytical approach for the NEP girders.

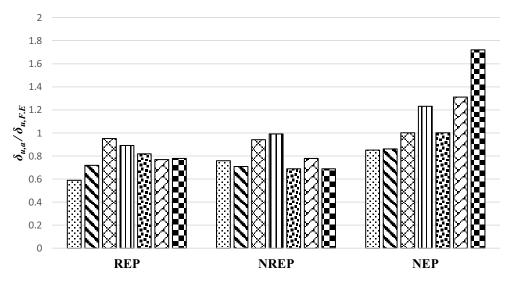
The same conclusions can be seen clearly form tables 5, 6 and 7, increasing the end-distance has insignificant effect on the deflection of the studied girders. Given that, for (e) equal 100, 150 and 300

13th International Conference on Civil and Architecture Engineering (ICCAE-13) IOP Publishing IOP Conf. Series: Materials Science and Engineering 974 (2020) 012012 doi:10.1088/1757-899X/974/1/012012

mm, the mean values of  $\delta_{pb,a'}/\delta_{pb,FE}$  are 0.84, 0.85 and 0.85 respectively for REP girders. And equals 0.86, 0.87 and 0.86 for NREP girders.



 $\square a/hw=0.8$   $\square a/hw=1$   $\square a/hw=1.4$   $\square a/hw=1.7$   $\square a/hw=2$   $\square a/hw=2.2$   $\square a/hw=3$ 



 $\square a/hw=0.8$   $\square a/hw=1$   $\square a/hw=1.4$   $\square a/hw=1.7$   $\square a/hw=2$   $\square a/hw=2.2$   $\square a/hw=3$ 

13th International Conference on Civil and Architecture Engineering (ICCAE-13)IOP PublishingIOP Conf. Series: Materials Science and Engineering 974 (2020) 012012doi:10.1088/1757-899X/974/1/012012

Figure 9. Comparison between the analytical and FE results (loads/deflections) with different end-post profiles.

	`	Analytical Eq.	FE (	FE (REP)	FE (NREP)	REP)	FE(NEP)	NEP)	Ratio	Ratio (REP)	Ratio (	Ratio (NREP)	Ratio	Ratio (NEP)
vd \r Cross Cross	ectio P <sub>u,a</sub>	$\delta_{pb,a}$	$P_{u,FE}$	$\delta_{pb,FE}$	$P_{u,FE}$	$\delta_{pb,FE}$	$P_{u,FE}$	$\delta_{pb,FE}$	$P_{\rm u,a}/P_{\rm u,FE}$	$\delta_{pb,a}/\delta_{pb,FE}$	$P_{u,d}\!\!\!/\!P_{u,FE}$	$\delta_{pb.a}/\delta_{pb.FE}$	$P_{u,a}/P_{u,FE}$	$\delta_{pb,a}/\delta_{pb,FE}$
	, kN	шш	kΝ	шш	kN	шш	kN	шш				l viod		
3.5	701	19.14	951	23.01	942	22.90	216	10.12	0.74	0.83	0.74	0.84	3.25	1.89
3.4	714	18.12	943	21.28	945	21.09	215	9.72	0.01	0.85	0.01	0.86	0.03	1.86
3.3	727	17.12	961	19.68	950	19.64	216	9.20	0.76	0.87	0.77	0.87	3.37	1.86
3.2	742	16.15	968	18.17	951	18.21	216	8.85	0.77	0.89	0.78	0.89	3.44	1.82
3.1 n	757	15.21	974	16.80	955	18.23	216	8.61	0.78	0.91	0.79	0.83	3.50	1.77
2 mu 3.0	773	14.30	626	18.34	961	20.62	216	8.33	0.79	0.78	0.80	0.69	3.58	1.72
2.8 2.8	808	12.57	166	15.20	996	13.00	216	7.51	0.82	0.83	0.84	0.97	3.74	1.67
= 300 5.6	848	10.95	866	12.18	971	11.40	217	7.18	0.85	06.0	0.87	0.96	3.91	1.53
2.4 4.1 ג לן	894	9.46	1004	12.64	980	12.80	217	6.67	0.89	0.75	0.91	0.74	4.12	1.42
, b <sub>3</sub>	947	8.08	1015	10.47	1001	10.36	216	6.17	0.93	0.77	0.95	0.78	4.38	1.31
2.0	1009	6.83	1037	8.28	1003	9.91	217	6.83	0.97	0.82	1.01	0.69	4.65	1.00
1.8 2 X	1081	5.69	1096	6.64	1046	7.00	216	5.13	0.99	0.86	1.03	0.81	5.00	1.11
1.7 1000	1123	5.16	1102	5.81	1065	5.20	202	4.18	1.02	0.89	1.05	66.0	5.56	1.23
1.6	1168	4.67	1097	5.00	1064	5.03	217	5.31	1.06	0.93	1.10	0.93	5.38	0.88
1.5 hw.	1217	4.20	1130	4.47	1082	4.52	272	1.93	1.08	0.94	1.12	0.93	4.47	2.18
1.4	1272	3.77	1155	3.98	1103	3.99	220	3.77	1.10	0.95	1.15	0.94	5.78	1.00
1.3	1332	3.36	1207	3.46	1130	3.47	225	3.78	1.10	0.97	1.18	0.97	5.92	0.89
1.2	1399	2.99	1218	4.42	1158	3.00	220	1.66	1.15	0.68	1.21	1.00	6.36	1.80
1.0	1556	2.33	1407	3.22	1237	3.28	230	2.70	1.11	0.72	1.26	0.71	6.77	0.86
0.8	1769	1.82	1596	3.10	1353	2.41	245	2.15	1.11	0.59	1.31	0.76	7.22	0.85
					M	Mean			0.94	0.84	80.0	0.86	4 60	1 43

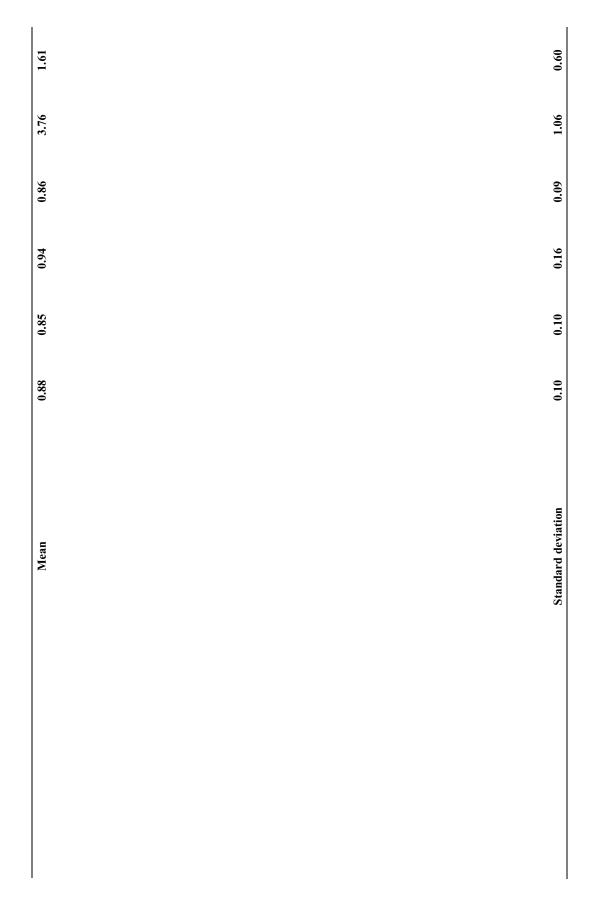
13th International Conference on Civil and Architecture Engineering (ICCAE	E-13) IOP Publishing
IOP Conf. Series: Materials Science and Engineering 974 (2020) 012012 do	bi:10.1088/1757-899X/974/1/012012

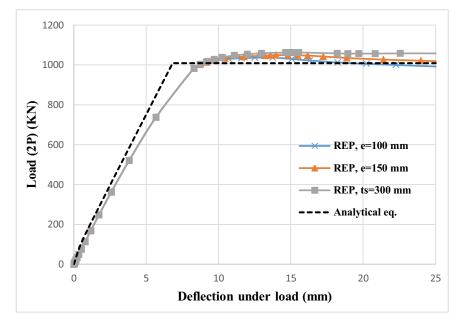
$= 10 \text{ mm, } c = 150 \text{ mm}$ Ratio (REP) Pu,FE $\delta_{pb,a}/\delta_{pb,FE}$ 74 0.83 75 0.87 75 0.87 75 0.87 76 0.89 77 0.91 79 0.69 81 0.83 84 0.91 79 0.69 88 0.97 96 0.82 97 0.85 97 0.93 96 0.95 01 0.88 03 0.93 05 0.94 06 0.97 06 1.00 04 0.71 06 1.00 04 0.71 03 0.57 01 0.85 01 0.85 01 0.85 01 0.85 01 0.85 02 0.95 03 0.95 04 0.71 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 04 0.71 03 0.57 03 0.57 03 0.57 03 0.57 03 0.57 04 0.71 03 0.57 05 0.55 05 0.55		i				D13	Standard devlation	evianon			<b>.</b>	01.0	QT.0	01.0	1.24	0.43
Analytical Eq         FE (NEF)         FE (NEF)         Ref (NEF)         Ratio (REP)           Fusa         δpa.         Pu.re         δpa.FE         Pu.re         δpa.FE         Pu.re         δpa.FE           (KN)         (mm)         (kN)         (mm)         (kN)         (mm)         (kN)         (mm)         (kN)           (Fu)         (mm)         (kN)         (mm)         (kN)         (mm)         (kN)         (mm)         (kN)           701         19.14         952         23.06         952         23.11         234         0.74         0.83           773         17.12         967         19.68         956         19.59         234         9.34         0.76         0.83           773         14.30         983         20.64         967         18.17         234         8.91         0.76         0.83           773         14.30         983         20.64         967         18.17         234         8.91         0.76         0.83           735         143         983         10.95         10.31         12.52         234         0.76         0.83           736         1936         11.4         53	Table	6. Compa	arison bet	ween ana	lytical a	nd FE lo	ıds/defl€	ctions rea	sults for	specime	(ts	nm, e=150 n	ım).			
Cross section         P <sub>1a</sub> $\delta_{pla}$ P <sub>u</sub> re $\delta_{pla}$ P <sub>u</sub> re $\delta_{pla}$ (KV)         (mm)         (KN)         (mm)         (KN)         (mm)         (KN)         (mm)         (KN)         (mm)         (KN)         (mm)         (KN)         (mm) $\delta_{pla}$			.Analy	rtical Eq	FE (	(REP)	FE (I	VREP)	FE(	NEP)	Ratio	(REP)	Ratio (	Ratio (NREP)	Ratio	Ratio (NEP)
section         (kN)         (k	мq	Cross	$\mathrm{P}_{\mathrm{u,a}}$	$\delta_{pb,a}$	$P_{u,\mathrm{FE}}$	$\delta_{pb,FE}$	$P_{u,\mathrm{FE}}$	$\delta_{pb,FE}$	$P_{u,\mathrm{FE}}$	$\delta_{pb,FE}$						
701         19.14         952         23.06         952         23.11         234         10.40         0.74         0.83           714         18.12         944         21.29         947         21.28         234         9.94         0.76         0.85           727         17.12         967         19.68         956         19.17         234         8.91         0.76         0.85           727         15.21         977         16.78         966         18.17         234         8.91         0.76         0.85           742         16.15         972         18.18         966         18.17         234         8.91         0.76         0.85           737         14.30         983         20.64         967         18.37         234         8.48         0.77         0.91           888         10.95         1004         11.29         986         11.46         2.23         6.96         0.87         0.75           947         1090         6.83         10.26         998         12.77         2.34         8.48         0.77         0.91           11009         6.83         10040         11.29         903         12	[ \ <b>B</b>	section	(kN)	(mm)	(kN)	(mm)	(kN)	(mm)	(kN)	(mm)	$P_{u,a'}P_{u,FE}$	$\delta_{pb,a}/\delta_{pb,FE}$	$P_{u,a}\!\!/P_{u,FE}$	$\delta_{pb,a}/\delta_{pb,FE}$	$P_{u,a}\!/P_{u,FE}$	$\delta_{pb,a}/\delta_{pb,FE}$
714         18.12         944         21.29         947         21.28         234         9.94         0.76         0.85           727         17.12         967         19.68         956         19.59         234         9.91         0.76         0.87           727         15.21         977         16.15         972         18.18         960         18.17         234         8.91         0.77         0.91           757         15.21         977         16.78         965         18.17         234         8.48         0.77         0.91           757         15.21         974         15.21         976         18.37         234         8.48         0.77         0.91           808         12.57         994         15.21         978         12.86         232         6.96         0.81         0.79         0.69           947         8.08         10.95         1004         1129         5.32         234         6.00         0.81         0.77         0.79         0.75           1009         6.83         1049         8.29         10.31         12.46         235         5.39         0.84         0.77         0.75      <	3.5		701	19.14	952	23.06	952	23.11	234	10.40	0.74	0.83	0.74	0.83	3.00	1.84
727         17.12         967         19.68         956         19.59         234         9.30         0.75         0.87           742         16.15         972         18.18         960         18.17         234         8.91         0.76         0.89           757         15.21         977         16.78         965         18.17         234         8.84         0.79         0.91           753         14.30         983         20.64         967         18.37         234         8.84         0.79         0.91           808         12.57         994         15.21         978         12.32         539         0.84         0.79         0.69           808         10.95         1004         11.29         986         11.46         235         5.39         0.84         0.77         0.91           947         8.08         10.95         1004         11.29         5.35         5.34         0.89         0.79         0.75           1009         6.83         1049         8.29         1025         1023         10.31         2.34         5.63         0.87         0.79         0.75           1100         6.83         104	3.4		714	18.12	944	21.29	947	21.28	234	9.94	0.76	0.85	0.75	0.85	3.05	1.82
742         16.15         972         18.18         960         18.17         234         8.91         0.76         0.89           757         15.21         977         16.78         965         18.17         234         8.84         0.77         0.91           773         14.30         983         20.64         967         18.37         234         8.48         0.79         0.69           808         12.57         994         15.21         978         12.86         232         6.96         0.81         0.63           804         916         1129         966         11.46         235         5.39         0.84         0.77         0.91           917         808         10.06         10.25         10.41         12.66         998         12.72         234         6.97         0.77         0.97           918         5.69         1114         6.68         1077         6.00         233         5.64         0.79         0.76           1031         5.69         1114         5.84         1077         6.00         233         5.66         0.87         0.76           1123         5.16         1114         5.84	3.3		727	17.12	967	19.68	956	19.59	234	9.30	0.75	0.87	0.76	0.87	3.11	1.84
751       15.21       977       16.78       965       18.17       234       8.84       0.77       0.91         773       14.30       983       20.64       967       18.37       234       8.48       0.79       0.69         808       12.57       994       15.21       978       12.86       232       6.96       0.81       0.69         848       10.95       1004       11.29       986       11.46       235       5.39       0.84       0.93         947       8.08       1060       10.25       1023       10.21       232       6.34       0.87       0.83         1009       6.83       1049       8.29       1073       12.72       234       6.57       0.87       0.75         1112       5.69       1114       5.68       1079       6.73       233       5.66       0.97       0.85         1123       5.16       1114       5.84       1077       6.00       234       9.95       1.01       0.85         1123       5.16       1114       5.84       1077       6.00       233       5.66       0.97       0.85         1232       3.36       1.34	3.2		742	16.15	972		960	18.17	234	8.91	0.76	0.89	0.77	0.89	3.17	1.81
773       14.30       983       20.64       967       18.37       234       8.48       0.79       0.69         808       12.57       994       15.21       978       12.86       232       6.96       0.81       0.83         848       10.95       1004       11.29       986       11.46       235       5.39       0.84       0.97         848       10.95       1004       11.29       986       11.46       235       5.39       0.84       0.97         947       808       1060       10.25       1023       10.31       235       6.34       0.89       0.79       0.75         1100       6.83       1049       8.29       1050       8.32       233       5.66       0.97       0.83         1112       5.16       1114       6.68       1077       6.00       234       10.97       0.88         1123       5.16       1114       5.84       1077       6.00       234       10.40       1.03       0.93         1133       3.10       3.18       4.07       2.33       5.66       0.97       0.83         1217       4.20       1115       4.48       2.34<	3.1	աս	757	15.21	777	16.78	965	18.17	234	8.84	0.77	0.91	0.78	0.84	3.24	1.72
$t_{i}$ 808         12.57         994         15.21         978         12.86         232         6.96         0.81         0.83 $t_{i}$ 848         10.95         1004         11.29         986         11.46         235         5.39         0.84         0.97 $t_{i}$ 848         10.95         1004         11.29         986         11.46         235         5.39         0.84         0.97 $t_{i}$ 894         9.46         1030         12.66         998         12.72         234         6.57         0.87         0.75 $t_{i}$ 1089         6.83         1049         8.29         1050         8.32         234         6.00         0.96         0.83 $t_{i}$ 1114         5.68         1077         6.00         234         9.95         1.01         0.83 $t_{i}$ 1118         5.64         1123         5.16         1114         5.84         1077         6.00         234         9.95         1.01         0.83 $t_{i}$ 1123         5.16         1114         5.84         1077         6.00         234         2	3.0	1 52	773	14.30	983	20.64	967	18.37	234	8.48	0.79	0.69	0.80	0.78	3.30	1.69
The first is the interval of t	2.8	x 00	808	12.57	994	15.21	978	12.86	232	6.96	0.81	0.83	0.83	0.98	3.48	1.81
$t_{\text{t}}$ 894       9.46       1030       12.66       998       12.72       234       6.57       0.87       0.75 $h_{\text{t}}$ 947       8.08       1060       10.25       1023       10.31       235       6.34       0.89       0.79       0.75 $5.5$ 1081       5.69       1114       6.82       1070       6.33       234       6.00       0.96       0.83       0.79       0.79 $5.5$ 1114       5.69       1114       6.68       1077       6.00       234       9.95       1.01       0.85 $1123$ 5.16       1114       5.84       1077       6.00       234       9.95       1.01       0.85 $1217$ 4.67       1129       5.00       1066       4.04       234       10.40       1.03       0.93 $1272$ 3.77       1205       3.98       1138       4.00       237       3.75       1.05       0.94 $1332$ 3.36       1205       3.47       1167       3.49       2.36       3.91       1.06       0.95 $1332$ 2.33       1332       3.34       1167       3.4	2.6	)£ =	848	10.95	1004	11.29	986	11.46	235	5.39	0.84	0.97	0.86	0.96	3.61	2.03
$h_{i}$ 947         8.08         1060         10.25         10.23         10.31         235         6.34         0.89         0.79 $h_{i}$ 1009         6.83         1049         8.29         1050         8.32         234         6.00         0.96         0.82 $h_{i}$ 1081         5.69         1114         6.68         1079         6.73         233         5.66         0.97         0.85 $h_{i}$ 1123         5.16         1114         5.84         1077         6.00         234         9.95         1.01         0.88 $h_{i}$ 1217         4.50         1157         4.48         1113         4.48         234         10.40         1.03         0.93 $h_{i}$ 1217         4.20         1157         4.48         1113         4.48         237         3.75         1.05         0.94 $1272$ 3.77         1205         3.98         1138         4.00         236         3.91         1.06         0.95 $1332$ 3.36         1259         3.47         1167         3.49         3.66         0.97         0.91	2.4	η χ	894	9.46	1030	12.66	866	12.72	234	6.57	0.87	0.75	06.0	0.74	3.82	1.44
Image <th< td=""><td>2.2</td><td>iq'ι</td><td>947</td><td>8.08</td><td>1060</td><td>10.25</td><td>1023</td><td>10.31</td><td>235</td><td>6.34</td><td>0.89</td><td>0.79</td><td>0.93</td><td>0.78</td><td>4.03</td><td>1.27</td></th<>	2.2	iq'ι	947	8.08	1060	10.25	1023	10.31	235	6.34	0.89	0.79	0.93	0.78	4.03	1.27
$5^{\circ}$ 1081       5.69       1114       6.68       1079       6.73       233       5.66       0.97 $1123$ 5.16       1114       5.84       1077       6.00       234       9.95       1.01 $1123$ 5.16       1114       5.84       1077       6.00       234       9.95       1.01 $1123$ 5.16       1114       5.84       1077       6.00       234       9.95       1.01 $1127$ 4.67       1129       5.00       1066       4.04       234       10.40       1.03 $1272$ 3.77       1205       3.98       1113       4.48       227       3.75       1.05 $1272$ 3.77       1205       3.98       1167       3.49       236       3.91       1.06 $1332$ 3.36       1257       3.47       1167       3.49       236       3.91       1.06 $1332$ 3.36       1202       3.02       243       3.34       1.06 $1339$ 2.99       1326       3.01       1202       3.02       243       3.61       1.06 $1556$ 2.33	2.0	шш	1009	6.83	1049	8.29	1050	8.32	234	6.00	0.96	0.82	0.96	0.82	4.31	1.14
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	1.8	ςx	1081	5.69	1114	6.68	1079	6.73	233	5.66	0.97	0.85	1.00	0.85	4.64	1.01
$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$	1.7	000	1123	5.16	1114	5.84	1077	6.00	234	9.95	1.01	0.88	1.04	0.86	4.80	0.52
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	1.6	I = '	1168	4.67	1129	5.00	1066	4.04	234	10.40	1.03	0.93	1.10	1.16	4.99	0.45
T       1272       3.77       1205       3.98       1138       4.00       230       3.78       1.06         1332       3.36       1259       3.47       1167       3.49       236       3.91       1.06         1332       3.36       1259       3.47       1167       3.49       236       3.91       1.06         1399       2.99       1326       3.00       1202       3.02       243       3.84       1.06         1556       2.33       1498       3.29       1297       3.31       236       3.72       1.04         1769       1.82       1723       3.17       1443       2.43       251       2.28       1.03	1.5	اw. t	1217	4.20	1157	4.48	1113	4.48	227	3.75	1.05	0.94	1.09	0.94	5.36	1.12
1332     3.36     1259     3.47     1167     3.49     236     3.91     1.06       1399     2.99     1326     3.00     1202     3.02     243     3.84     1.06       1556     2.33     1498     3.29     1297     3.31     236     3.72     1.04       1769     1.82     1723     3.17     1443     2.43     251     2.28     1.03	1.4	4	1272	3.77	1205	3.98	1138	4.00	230	3.78	1.06	0.95	1.12	0.94	5.53	1.00
1399     2.99     1326     3.00     1202     3.02     243     3.84     1.06       1556     2.33     1498     3.29     1297     3.31     236     3.72     1.04       1769     1.82     1723     3.17     1443     2.43     251     2.28     1.03	1.3		1332	3.36	1259	3.47	1167	3.49	236	3.91	1.06	0.97	1.14	0.96	5.64	0.86
1556     2.33     1498     3.29     1297     3.31     236     3.72     1.04       1769     1.82     1773     3.17     1443     2.43     251     2.28     1.03       Mean	1.2		1399	2.99	1326	3.00	1202	3.02	243	3.84	1.06	1.00	1.16	0.99	5.76	0.78
1769         1.82         1723         3.17         1443         2.43         2.51         2.28         1.03           Mean         Mean         0.91	1.0		1556	2.33	1498	3.29	1297	3.31	236	3.72	1.04	0.71	1.20	0.70	6.59	0.63
0.91	0.8		1769	1.82	1723	3.17	1443	2.43	251	2.28	1.03	0.57	1.23	0.75	7.05	0.80
							M	ean			0.91	0.85	96.0	0.87	4.42	1.28

13th International Conference on Civil and Architecture Engineering (ICCAE-13)IOP PublishingIOP Conf. Series: Materials Science and Engineering 974 (2020) 012012doi:10.1088/1757-899X/974/1/012012

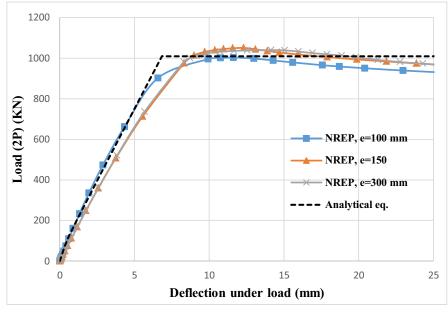
able 7. C	Compar	150n between a1 .Analytical Eq	veen ana ical Eq	Table 7. Comparison between analytical and FE         .Analytical Eq       FE (REP)		ads/defle FE (N	(deflections res FE (NREP)	sults for FE(1	s for specime FE(NEP)	$\frac{\text{ns (ts = 10 1)}}{\text{Ratio}}$	loads/deflections results for specimens (ts = 10 mm, e =300 mm) FE (NREP) FE(NEP) Ratio (REP)		Ratio (NREP)	Ratio	Ratio (NEP)
	Cross	$P_{u,a}$	$\delta_{pb,a}$	$\mathbf{P}_{\mathrm{u,FE}}$	$\delta_{pb,FE}$	$\mathbf{P}_{\mathrm{u,FE}}$	$\delta_{pb,FE}$	$\mathbf{P}_{\mathrm{u,FE}}$	$\delta_{pb,FE}$						
a/ ]	section	(kN)	(mm)	(kN)	(mm)	(kN)	(mm)	(kN)	(mm)	$P_{u,a\!/}P_{u,FE}$	$\delta_{pb,a}/\delta_{pb,FE}$	$P_{u,a}\!/P_{u,FE}$	$\delta_{pb,a'} \delta_{pb,FE}$	$P_{u,a}\!\!/\!P_{u,FE}$	$\delta_{pb,a}/\delta_{pb,FE}$
3.5		701	19.14	955	23.09	953	23.09	275	9.21	0.73	0.83	0.74	0.83	2.55	2.08
3.4		714	18.12	946	21.30	947	21.29	276	8.84	0.75	0.85	0.75	0.85	2.59	2.05
3.3		727	17.12	970	19.68	965	19.66	276	8.10	0.75	0.87	0.75	0.87	2.63	2.11
3.2		742	16.15	975	18.18	972	18.17	275	7.69	0.76	0.89	0.76	0.89	2.70	2.10
3.1	աս	757	15.21	982	16.78	978	18.16	275	8.40	0.77	0.91	0.77	0.84	2.75	1.81
3.0	1 52	773	14.30	987	20.06	981	20.65	276	6.75	0.78	0.71	0.79	0.69	2.80	2.12
2.8	x 0	808	12.57	866	19.43	992	12.86	276	5.86	0.81	0.65	0.81	0.98	2.93	2.15
2.6	)E =	848	10.95	1014	11.31	1001	11.37	275	5.17	0.84	0.97	0.85	0.96	3.08	2.12
2.4	h x	894	9.46	1044	12.68	1020	12.69	276	4.62	0.86	0.75	0.88	0.75	3.24	2.05
2.2	₁d ,	947	8.08	1075	10.28	1046	10.29	276	4.04	0.88	0.79	0.91	0.79	3.43	2.00
2.0	աա	1009	6.83	1062	8.30	1039	8.29	270	3.14	0.95	0.82	0.97	0.82	3.74	2.18
1.8	ç	1081	5.69	1133	6.69	1105	6.70	275	3.35	0.95	0.85	0.98	0.85	3.93	1.70
1.7	¢ 00	1123	5.16	1153	5.85	1095	5.74	276	8.64	0.97	0.88	1.03	0.90	4.07	0.60
1.6	01 =	1168	4.67	1191	5.01	1110	5.00	275	9.21	0.98	0.93	1.05	0.93	4.25	0.51
1.5	wj.	1217	4.20	1240	4.50	1140	4.48	273	3.04	0.98	0.93	1.07	0.94	4.46	1.38
1.4	"ч	1272	3.77	1299	4.00	1169	4.00	277	2.72	0.98	0.94	1.09	0.94	4.59	1.39
1.3		1332	3.36	1366	3.50	1201	3.49	277	2.77	0.98	0.96	1.11	0.96	4.81	1.21
1.2		1399	2.99	1449	3.03	1237	3.03	278	2.68	0.97	0.99	1.13	0.99	5.03	1.12
1.0		1556	2.33	1647	3.36	1338	3.29	282	3.37	0.94	0.69	1.16	0.71	5.52	0.69
0.8		1760	001	1000					!	000					

13th International Conference on Civil and Architecture Engineering (ICCAE-13)IOP PublishingIOP Conf. Series: Materials Science and Engineering 974 (2020) 012012doi:10.1088/1757-899X/974/1/012012

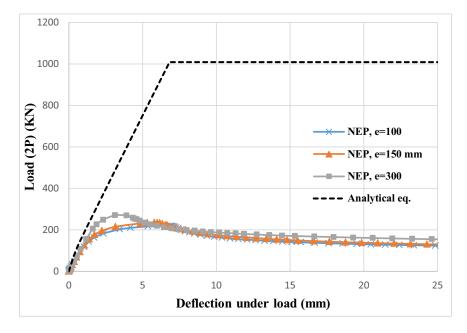




(a) REP



(b) NREP



(a) NEP

Figure 10. Analytical and FE load-deflection curves for plate girder  $(a/h_w = 2.0)$  with different enddistances.

# 5. CONCLUSIONS

The load/deflection of thin-walled panel loaded in shear were studied using numerical and analytical methods. The Qian and Tan [25] mechanical approach, to calculate the in-plane deflection web panel loaded in shear, has been applied to calculate pre- and post-buckling deflection. And hence the non-linear large deflection FE analysis of the full-scale plate girders were studied. The comparison between the results of two methods concluded that:

- The analytical approach presented by Qian and Tan [25] can be used to simulate the load/deflection behaviour of thin-walled panel loaded in shear for  $a/h_w < 2.2$ , with REP and NREP. Such that, it estimates well the shear strength of REP plate girders and slightly overestimates the shear strength for NREP plate girders.
- The presented analytical approach can't be used for calculating the deflection of NEP plate girders. And more studied is needed to find an accurate approach for NEP panels.
- The stiffeners thicknesses have insignificant effect on the load-deflection behaviour.
- The type of end-post directly affects the ultimate shear strength  $(P_u)$ . The results show that providing a REP increases the ultimate shear strength  $(P_{u,FE})$  of the girder, if compared with NREP.
- Changing end-post from REP to NREP has insignificant effect on maximum vertical deflection  $(\delta_{Pb,FE})$ .
- The FE results show that increasing the end-distance (e) has no effect on the initial stiffness of the plate girders. However, it slightly increases the ultimate shear strength of the girders.

#### **Conflict** of interest

On behalf of all authors, the corresponding author states that there is no conflict of interest.

#### 6. **REFERENCES**

[1] M. Hirt, J.-P. Lebet, Steel Bridges: Conceptual and Structural Design of Steel and Steel-Concrete Composite Bridges, EPFL Press 2013.

[2] N.M. Yossef, S.Taher, Cost Optimization of Composite Floor Systems with Castellated Steel Beams, Practice Periodical on Structural Design and Construction 24(1) (2018).

[3] S.C. Lee, J.S. Davidson, C.H. Yoo, Shear buckling coefficients of plate girder web panels, Computers & Structures 59(5) (1995) 789-795.

[4] S.C. Lee, C.H. Yoo, Strength of Plate Girder Web Panels under Pure Shear, Journal of Structural Engineering 124(2) (1998).

[5] C.H. Yoo, S.C. Lee, Mechanics of Web Panel Postbuckling Behavior in Shear Journal of Structural Engineering 132(10) (2006).

[6] M.M. Alinia, H.R. Habashi, A. Khorram, Nonlinearity in the postbuckling behaviour of thin steel shear panels, Thin-Walled Struct. 47(4) (2009) 412-420.

[7] M.M. Alinia, M. Shakiba, H.R. Habashi, Shear failure characteristics of steel plate girders, Thin-Walled Struct. 47(12) (2009) 1498-1506.

[8] J. Loughlan, N. Hussain, The in-plane shear failure of transversely stiffened thin plates, Thin-Walled Struct. 81 (2014) 225-235.

[9] G. Quan, S.-S. Huang, I. Burgess, An analytical approach to modelling shear panels in steel beams at elevated temperatures, Eng. Struct. 85(15) (2015) 73-82.

[10] Y.B. Kwon, S.W. Ryu, The shear strength of end web panels of plate girders with tension field action, Thin-Walled Struct. 98(January) (2016) 578-591.

[11] Y. Xiao, X.Y. Xue, F. FeiSun, G.Q. Li, Postbuckling shear capacity of high-strength steel plate girders, Journal of Constructional steel research 150(November) (2018) 475-490.

[12] A. Reis, N. Lopes, P.V. Real, Ultimate shear strength of steel plate girders at normal and fire conditions, Thin-Walled Struct. 137(April) (2019) 318-330.

[13] N.M. Yossef, J. Fogarty, S. El-Tawil, Residual Compressive Strength of Flange-Damaged Steel Columns, Journal of Structural Engineering (2015).

[14] O.F. Zaher, N.M. Yossef, M.H. El-Boghdadi, M.A. Dabaon, Structural behaviour of arched steel beams with cellular openings, Journal of Constructional steel research 148 (2018) 756-767.

[15] D.W. White, M.G. Barker, Shear Resistance of Transversely Stiffened Steel I-Girders, Journal of Structural Engineering 134 (2008) 1425-1436.

[16] D.M. Porter, K.C. Rockey, H.R. Evans, The collapse behaviour of plate girders loaded in shear, The Structural Engineer 53(8) (1975) 313 - 325.

[17] W. Ajam, C. Marsh, Simple model for shear capacity of webs, Journal of Structural Engineering 117(2) (1991) 410-422.

[18] T.V. Galambos, Guide to stability design criteria for metal structures, 5th ed., McGraw-Hill, New York., 1998.

[19] J. Wilson, On specifications for strength of iron bridges, Trans. Am. Soc. Civ. Eng. 15(1) (1886) 489-490.

[20] K. Basler, B. Thurlimann, Plate Girder Research, AISC Nat. Engrg. Conf., American Institute of Steel Construction, New York, N.Y., 1959.

[21] K. Basler, Strength of plate girders under combined bending and shear., Journal of the Structural Division, ASCE 87(2) (1961) 97-181.

[22] K. Basler, Strength of plate girders in shear, Transaction of ASCE, American Society of Civil Engineers., New York, 1961.

[23] K.C. Rocky, M. Skaloud, The ultimate load behaviour of plate girders loaded in shear, The Structural Engineer 50(1) (1972) 29 - 47.

[24] T. Höglund, Strength of steel and aluminum plate girders shear buckling and overall buckling of plane and trapezoidal webs comparisons with test, Dept. of Structural Engineering, Royal Institute of Technology, Stockholm, Sweden., 1995.

[25] Z.-H. Qian, K.-H. Tan, Deflection behaviour of plate girders loaded in shear at elevated temperatures, Journal of Constructional steel research 65 (2009) 991 - 1000.

[26] Y. Zhu, J. Zhao, Experimental and numerical study on member classification of I-section plate girders, Journal of Constructional steel research 137(October) (2017) 192-200.

[27] K.C. Rockey, E. H.R., D.M. Porter, A design method for predicting the collapse behavior of plate girders Proceedings of the Institution of Civil Engineers 65(1) (1978) 85-112.

[28] ABAQUS Standard User's Manual, The Abaqus Software is a product of Dassault Systèmes Simulia Corp, DassaultSystèmes, Providence, RI, USA, 2008.

[29] CEN, EN 1993-1-5, Eurocode 3. Design of steel structures - Part 1-5, Plated structural elements., 2007.

Nomer	nclature
$\sigma_t^y$	tensile yield stress of web panel under shear;
a	panel length;
$a/h_w$	web panel aspect ratio;
$A_w$	area of the web;
$b_f$	width of the flange;
С	distance assumed between the plastic hinge and the end stiffener;
Ε	elastic modulus;
$f_{yf}$	yield stress of the flange plate;
$f_{yw}$	yield stress of the web plate;
$h_w$	height of the web;
$h_w/t_w$	web slenderness ratio;
K	shear buckling coefficient;
L	span of the plate girder;
$M_{pf}$	plastic moment capacity of flanges $(= 0.25 f_{yf} A_f t_f)$ ;
$P_{u,a}$	ultimate shear force
$t_f$	thickness of the flange;
$t_s$	thickness of the stiffeners;
$t_w$	thickness of the web;
$V_{ult}$	post-buckling shear force component;
Wo	initial imperfection
$\delta_b$	deflection of web panel from pre-buckling to the post buckling stage;
$\delta_{cr}$	deflection of web panel at the pre-buckling stage;
$\delta_{pb}$	total deflection of web at the post buckling stage for a panel loaded in shear;
v	Poisson's ratio;
$ au_{cr}$	shear buckling stress of the web panel
$\theta$	inclination of the tensile field;